

Counterflow Combustion with Multiple Flames under High Strain Rates

William A. Sirignano
University of California, Irvine
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- Multi-branched flames are commonly found in unsteady combustors.
- They especially can appear with the extinction/ re-ignition behavior in vorticity-dominated flows with time-varying rates of strain S .
- Flamelet theory provides a useful model with reduced computational cost for multidimensional combustor analysis via CFD.
- Flamelet theory is based on counterflow similarity solutions in planar or axisymmetric configurations. *Peters (2000), Pierce & Moin (2004)*
- The theory has been developed only for single nonpremixed or premixed flames.
- Here, we extend the flamelet theory to three-dimensional configurations with multiple (one, two, or three) flames.

Governing Equations for 3D Reacting Counterflow

$$\frac{\partial \rho}{\partial t} + \frac{\partial(\rho u_j)}{\partial x_j} = 0 \quad \rho \frac{\partial Y_m}{\partial t} + \rho u_j \frac{\partial Y_m}{\partial x_j} = \frac{\partial}{\partial x_j} \left(\rho D \frac{\partial Y_m}{\partial x_j} \right) + \rho \omega_m \quad ; \quad m = 1, 2, \dots, N$$

$$\rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} + \frac{\partial p}{\partial x_i} = \frac{\partial \tau_{ij}}{\partial x_j} \quad \tau_{ij} = \mu \left[\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \frac{\partial u_k}{\partial x_k} \right]$$

$$\rho \frac{\partial h}{\partial t} + \rho u_j \frac{\partial h}{\partial x_j} - \frac{\partial p}{\partial t} = \frac{\partial}{\partial x_j} \left(\rho D \frac{\partial h}{\partial x_j} \right) - \rho \sum_{m=1}^N h_{f,m} \omega_m \quad Le = 1 \quad ; \quad M \ll 1$$

$$\eta \equiv \int_0^y \bar{\rho}(y') dy' \quad \text{for variable density flow.}$$

Similarity form

$$\rho = \rho(\eta) \quad ; \quad h = h(\eta) \quad ; \quad Y_m = Y_m(\eta)$$

$$\rho v = -S_1 f_1(\eta) - S_2 f_2(\eta) \quad ; \quad u = S_1 x (df_1/d\eta) \quad ; \quad w = S_2 z (df_2/d\eta).$$

ODEs in Similarity Analysis

Perfect gas , constant specific heats

$$f_1''' + (S_1 f_1 + S_2 f_2) f_1'' + S_1 (\tilde{h} - (f_1')^2) = 0$$

$$f_2''' + (S_1 f_1 + S_2 f_2) f_2'' + S_2 (\tilde{h} - (f_2')^2) = 0$$

$$f_1'(\infty) = \sqrt{\rho_{-\infty}} f_1'(-\infty) = f_2'(\infty) = \sqrt{\rho_{-\infty}} f_2'(-\infty) = 1$$

$$f_1(0) = f_2(0) = 0$$

$$Y_F'' + Pr(S_1 f_1 + S_2 f_2) Y_F' = Pr \omega_F$$

$$Y_O'' + Pr(S_1 f_1 + S_2 f_2) Y_O' = \nu Pr \omega_F$$

$$Y_F(\infty) = Y_{F,\infty} ; Y_F(-\infty) = Y_{F,-\infty}$$

$$\tilde{h}'' + Pr(S_1 f_1 + S_2 f_2) \tilde{h}' = Pr \tilde{Q} \omega_F^{\circ}$$

$$\tilde{h}(\infty) = 1 ; \tilde{h}(-\infty) = \frac{1}{\rho_{-\infty}}$$

One-step Westbrook-Dryer kinetics for propane and oxygen

$$\frac{dY_F}{dt} = \omega_F = -\frac{A^* \rho_\infty^{*0.75}}{S_1^* + S_2^*} \tilde{h}^{-0.75} Y_F^{0.1} Y_O^{1.65} e^{-50.237/\tilde{h}}$$
$$\omega_F = -\frac{Da}{\tilde{h}^{0.75}} Y_F^{0.1} Y_O^{1.65} e^{-50.237/\tilde{h}}$$

$$A^* = 4.79 \times 10^8 (\text{kg/m}^3)^{-0.75}$$

$$T_{\text{ambient}} = 300 \text{ K.}$$

$$\text{Reference Values: Strain Rate } S^* = S1^* + S2^* = 100/\text{s}$$

$$\text{Ambient Density } \rho^* = 10 \text{ kg/m}^3$$

$$Da = K Da_{\text{ref}}$$

$$Da_{\text{ref}} \equiv \frac{\tilde{A} (10 \text{ kg/m}^3)^{0.75}}{(100/\text{s})} = 2.693 \times 10^7 ; K \equiv \left[\frac{\rho_\infty^*}{10 \text{ kg/m}^3} \right]^{0.75} \frac{100/\text{s}}{S_1^* + S_2^*}$$

***K* increases with increasing pressure
and decreases with increasing rate of strain**

Conserved Scalars

$$\alpha \equiv Y_F - \nu Y_O$$

$$\beta \equiv \tilde{h} + \nu Y_O \tilde{Q}$$

$$\nu = 0.275$$

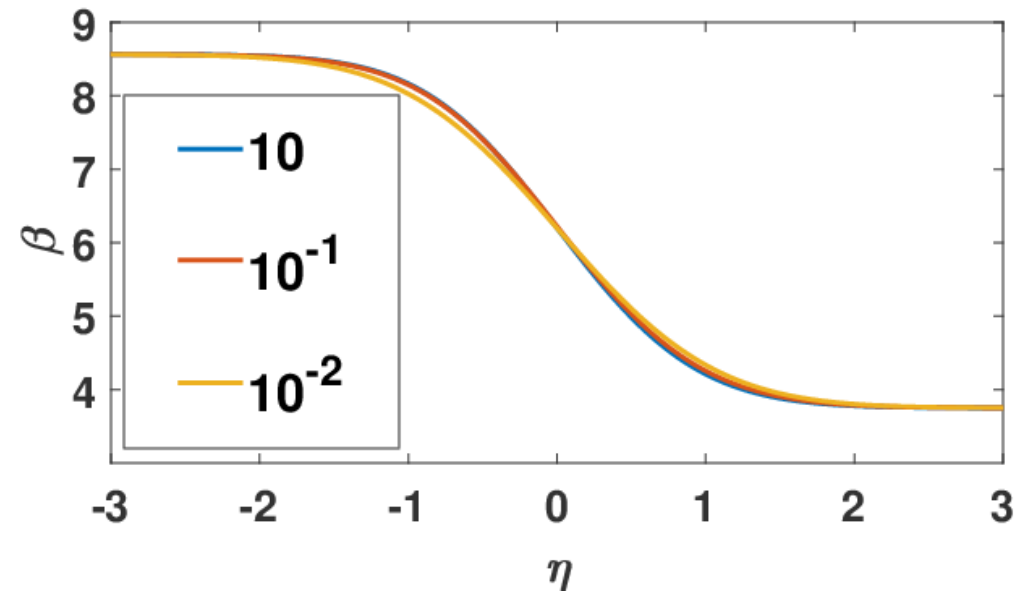
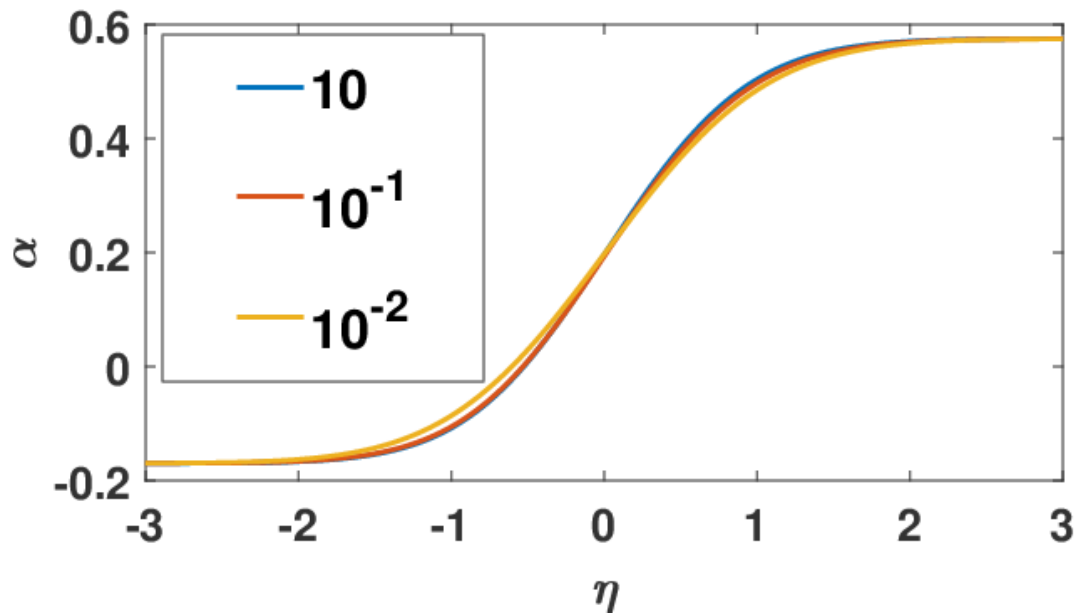
First case

A fuel-lean mixture flows from left to right.

A fuel-rich mixture flows from right to left.

$S_1 = 0.25$; $S_2 = 0.75$; $Pr = Sc = 1.0$; K given in Legend

Monotonic behavior, little effect of strain distribution

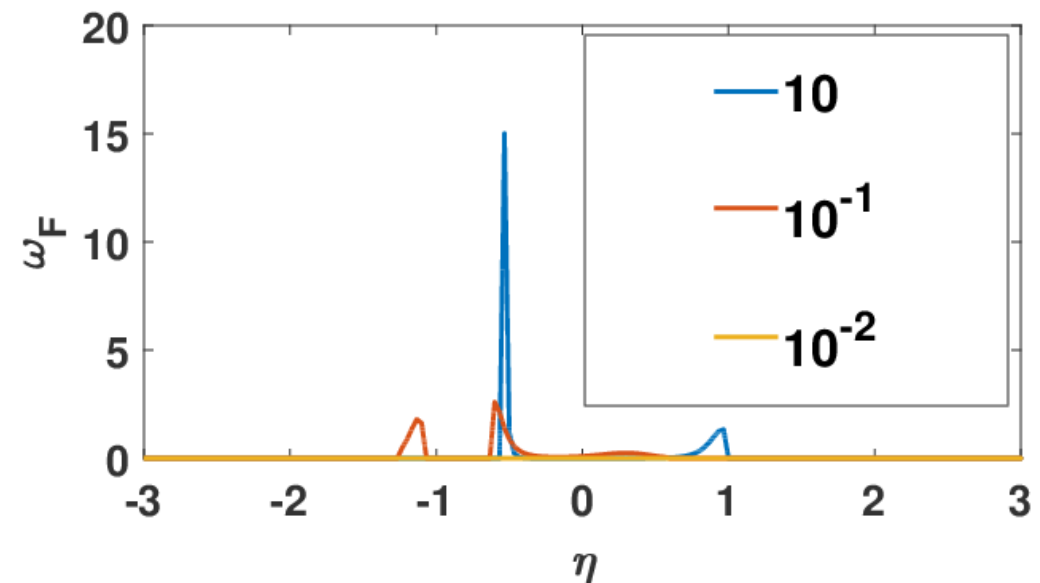
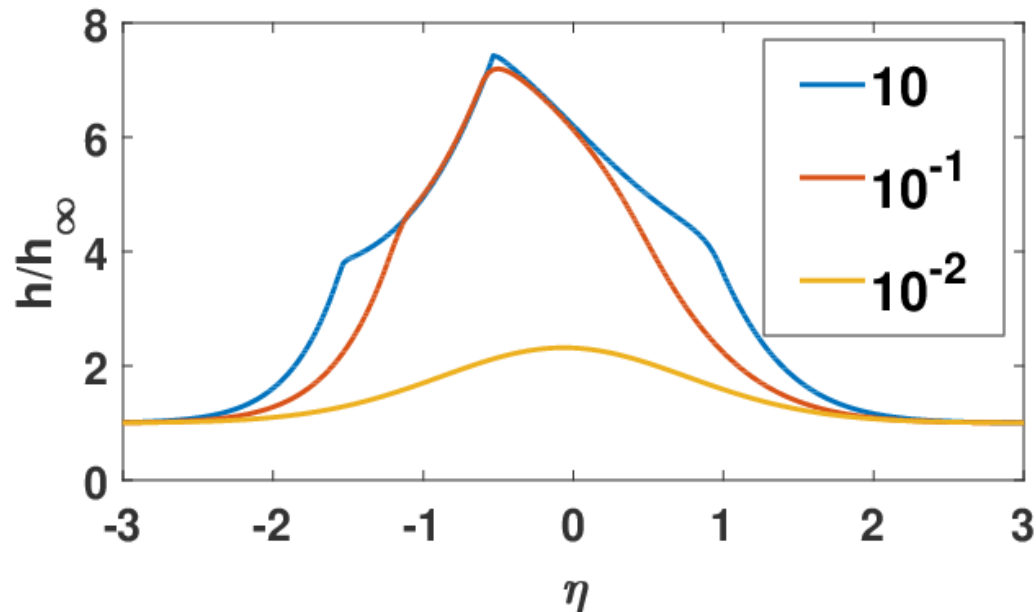


$$S_1 = 0.25 \quad ; \quad S_2 = 0.75 \quad ; \quad Pr = Sc = 1.0$$

-- **Three flames** can appear; fuel-lean premixed flame on left, diffusion flame in the middle, and fuel-rich premixed flame on right.

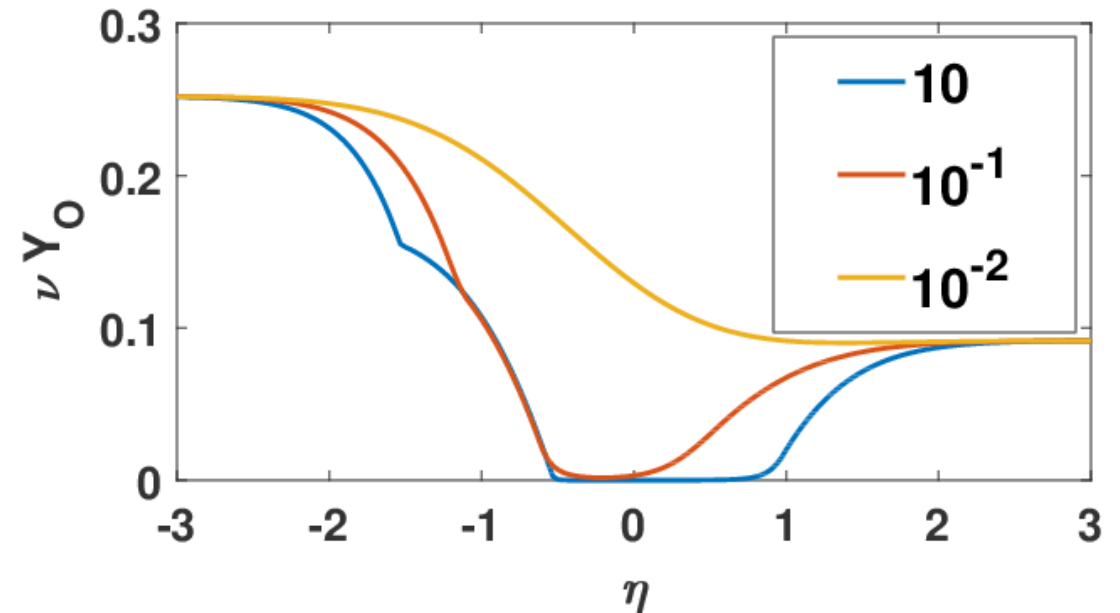
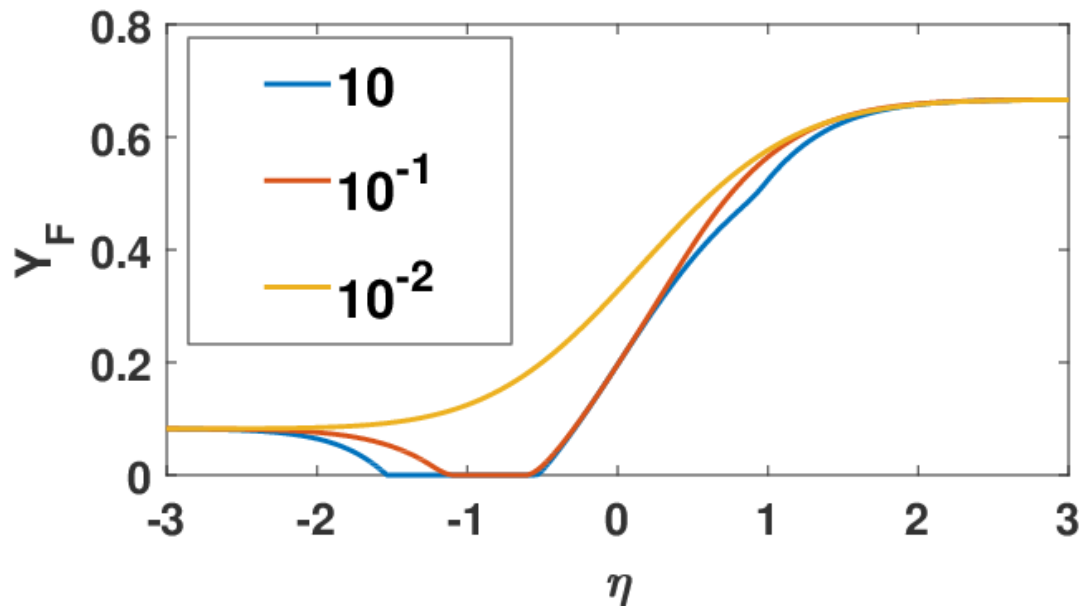
-- Increase in strain rate and/or decrease in pressure causes fuel-rich premixed flame to merge into diffusion flame.

-- Further increase in strain rate or decrease in pressure causes fuel-lean flame to merge with diffusion flame and then extinction with further change.



$$S_1 = 0.25 \quad ; \quad S_2 = 0.75 \quad ; \quad Pr = Sc = 1.0$$

- A domain with no fuel but substantial oxygen exists between the fuel-lean premixed flame and the diffusion flame.
- A domain with no oxygen but substantial fuel exists between the fuel-rich premixed flame and the diffusion flame.
- The diffusion flame sits in the fuel-lean stream.
- Merger and extinction are again shown with increasing strain rate and decreasing pressure.

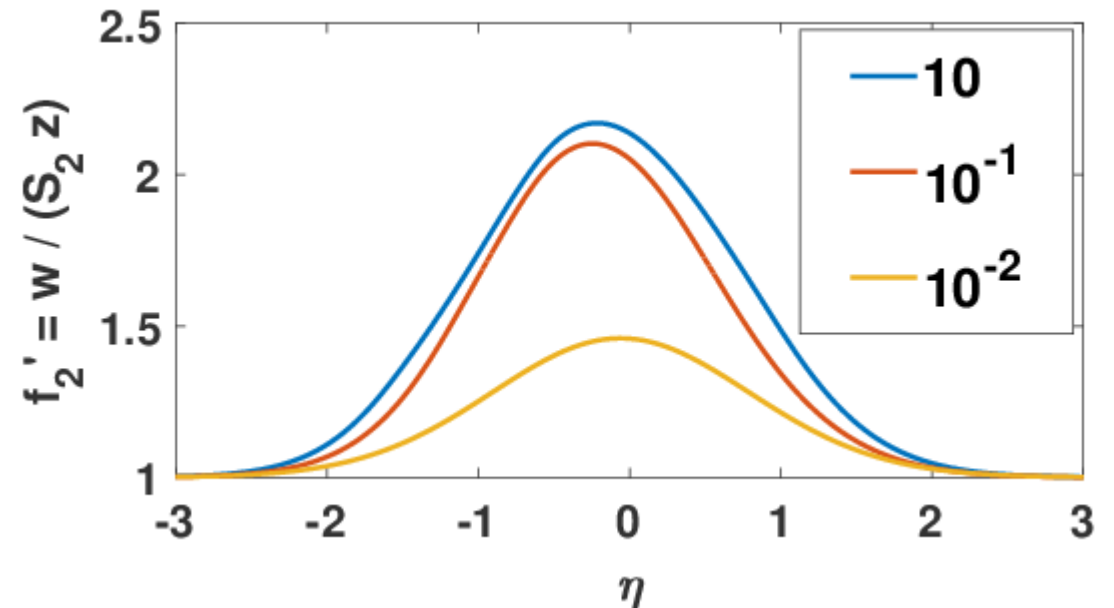
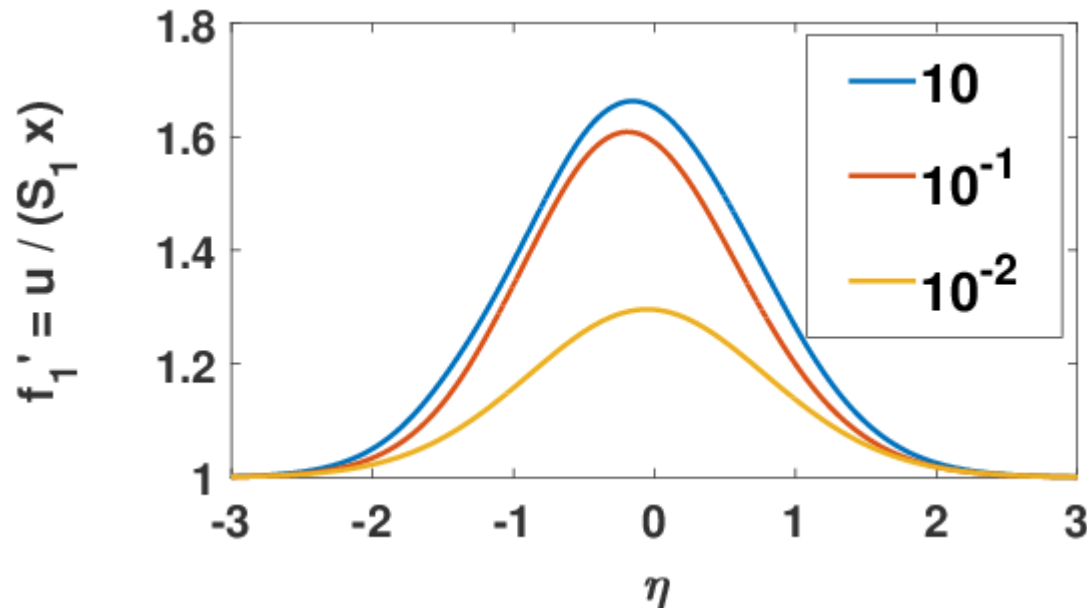


$$S_1 = 0.25 \quad ; \quad S_2 = 0.75 \quad ; \quad Pr = Sc = 1.0$$

-- In counterflows, the fluid in both streams is accelerated in the transverse directions away from the two symmetry planes. $\partial p / \partial x \sim -x$; $\partial p / \partial z \sim -z$; $u \sim x$; $w \sim z$.

-- The flames result in domains of high temperature and low density which have greater acceleration due to the pressure gradient.

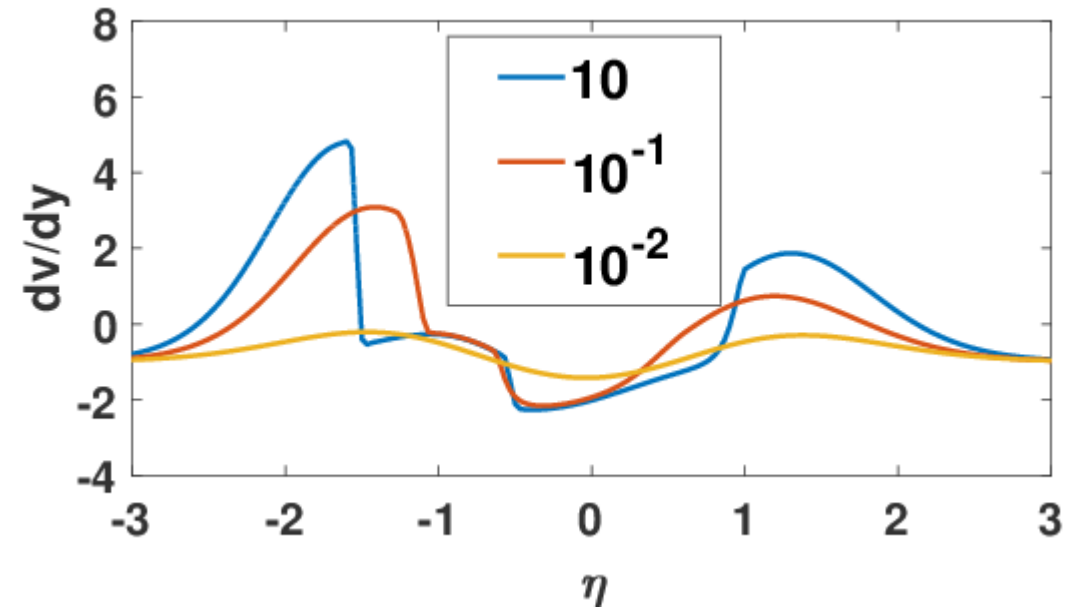
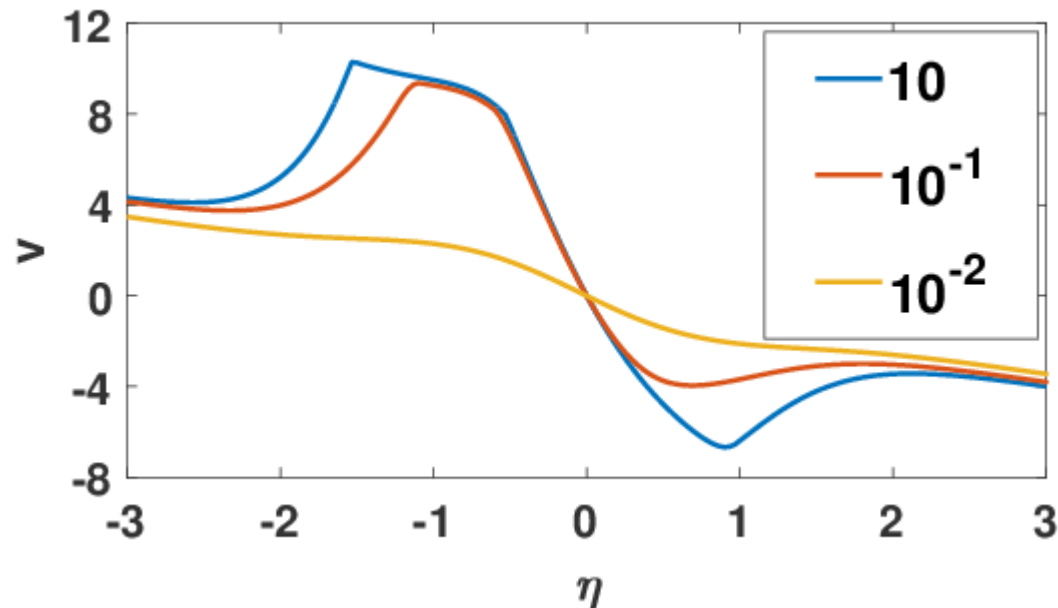
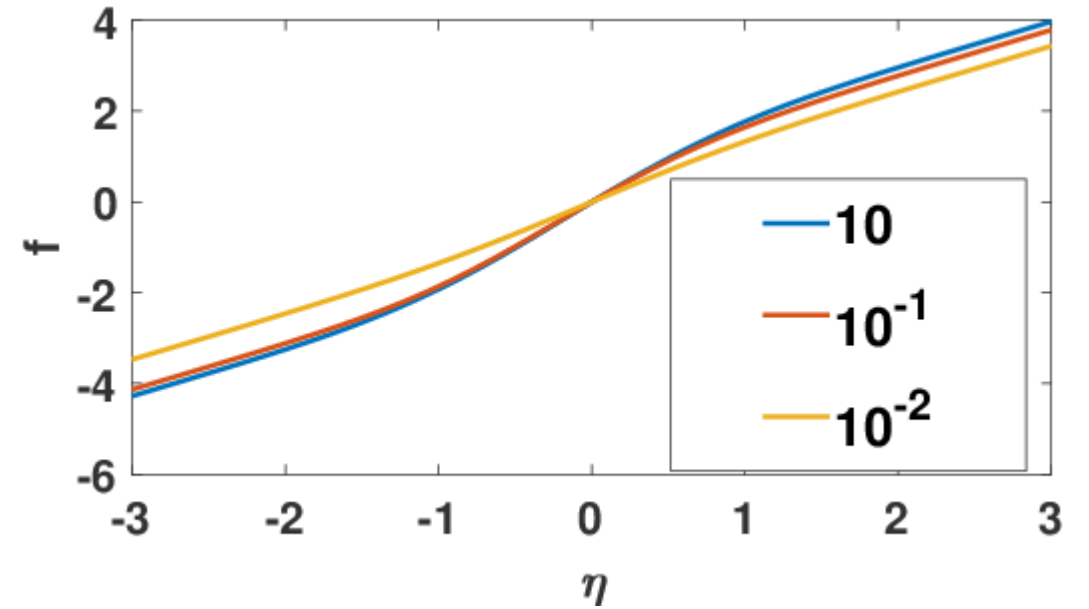
-- Overshoot of the transverse velocity occurs with greater velocities in the direction with greater normal strain rate.



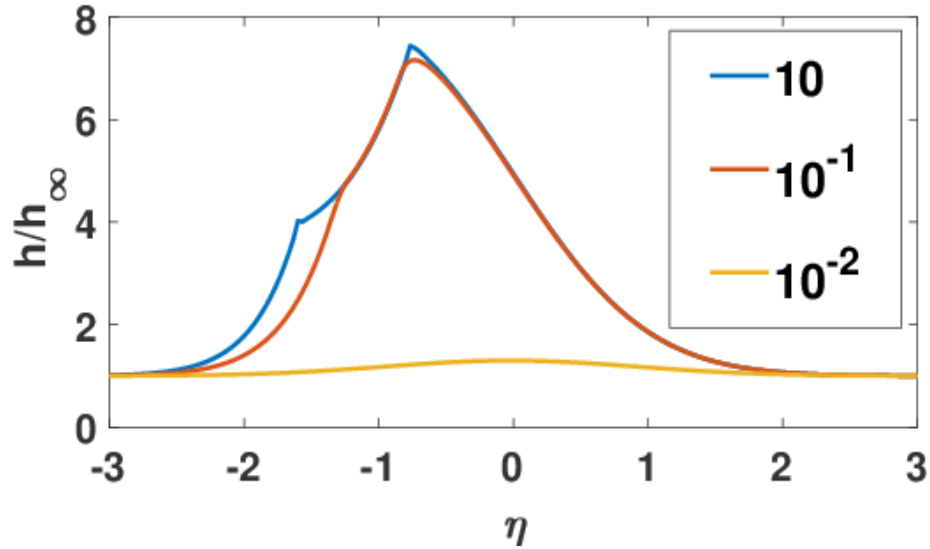
$$S_1 = 0.25 \quad ; \quad S_2 = 0.75 \quad ; \quad Pr = Sc = 1.0$$

Contrary to impressions given in the literature, the velocity v in the counterflow direction is far from linear (or even monotonic) in y .

There can be local maxima and minima in both The velocity v and the normal strain rate $\partial v / \partial y$.



Fuel-lean Mixture Flowing Against Fuel



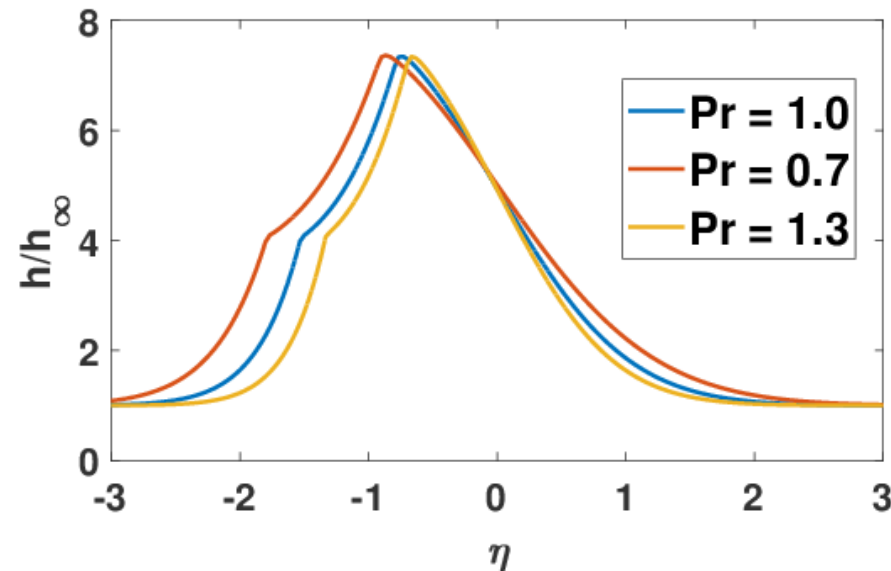
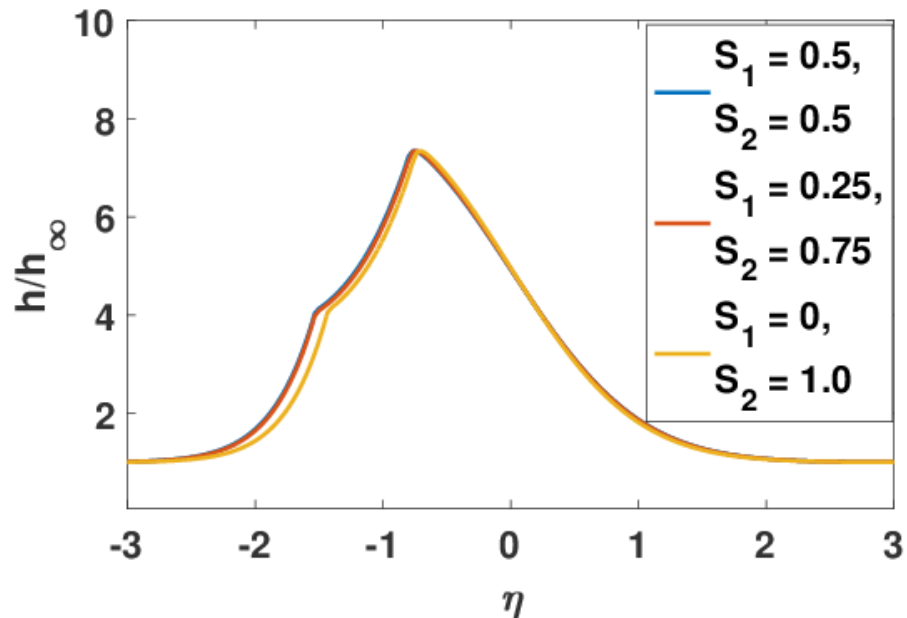
-- Here, results are presented for pure fuel flowing from the right counter to a fuel-lean combustible mixture from the left.

-- Only a fuel-lean premixed flame and a diffusion flame can occur.

-- Merging and extinction can follow as before with decreasing K value.

-- Strain rate distribution has little effect on scalar Properties; 3D, axisymmetric, and planar results are close.

-- Prandtl number has a more significant effect.



A new conserved scalar Σ must replace mixture fraction Z . It need not be physically meaningful but only monotonic in the y coordinate.

$$\Sigma'' + Pr(S_1 f_1 + S_2 f_2) \Sigma' = 0$$

$$\Sigma(\infty) = 1 ; \quad \Sigma(-\infty) = 0$$

$$\Sigma = \frac{\alpha(\eta) - \alpha(-\infty)}{\alpha(\infty) - \alpha(-\infty)} = \frac{\beta(\eta) - \beta(-\infty)}{\beta(\infty) - \beta(-\infty)}$$

$$\Sigma(\eta) = \frac{J(\eta)}{J(\infty)}$$

$$J(\eta) \equiv \int_{-\infty}^{\eta} e^{-I(\eta')} d\eta'$$

$$I(\eta) \equiv \int_{-\infty}^{\eta} Pr [S_1 f_1(\zeta) + S_2 f_2(\zeta)] d\zeta$$

$$2\chi \frac{d^2 Y_m}{d\Sigma^2} + Pr \omega_m = 0 ; \quad m = 1, 2, \dots, N$$

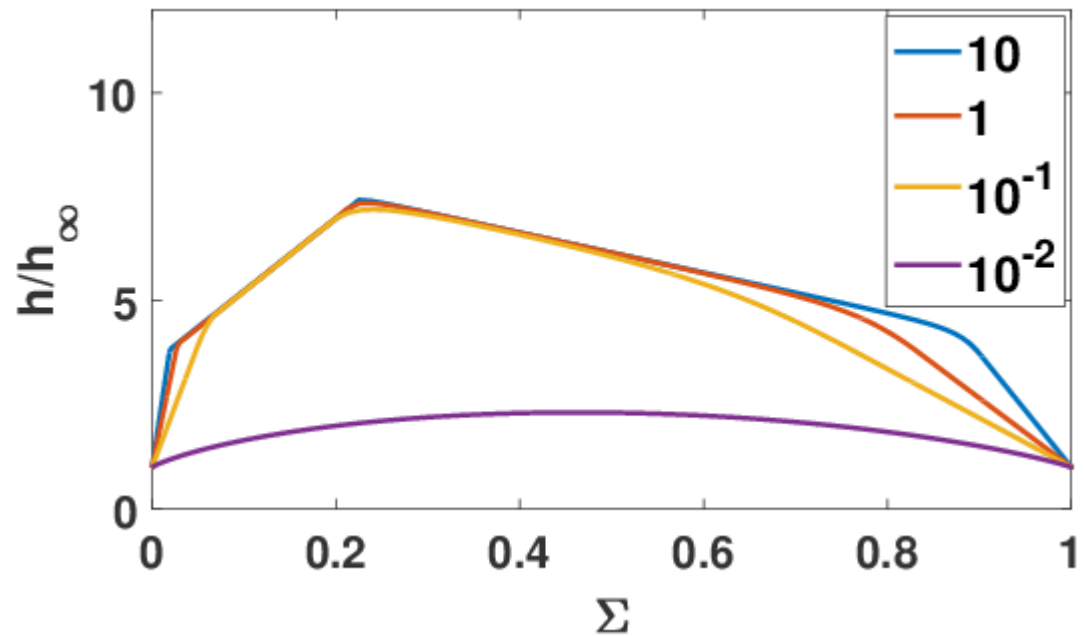
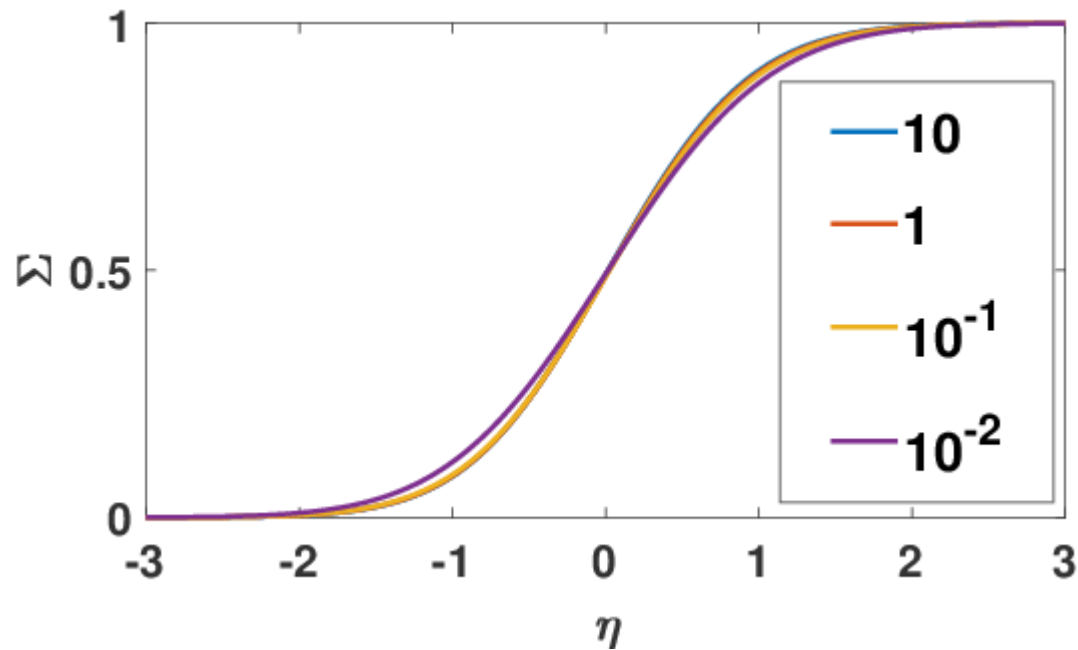
$$2\chi \frac{d^2 \tilde{h}}{d\Sigma^2} + Pr \tilde{Q} \omega_m = 0$$

$$\chi \equiv \frac{1}{2} \left(\frac{d\Sigma}{d\eta} \right)^2 = \frac{1}{2\rho^2} \left(\frac{d\Sigma}{dy} \right)^2 = \frac{1}{2} \frac{e^{-2I(\eta)}}{J^2(\infty)}$$

-- Σ is always a monotonic function of y or η while Z based on molecular species is not for multi-flame configuration. Z based on atomic species need not be monotonic in y for the unsteady state.

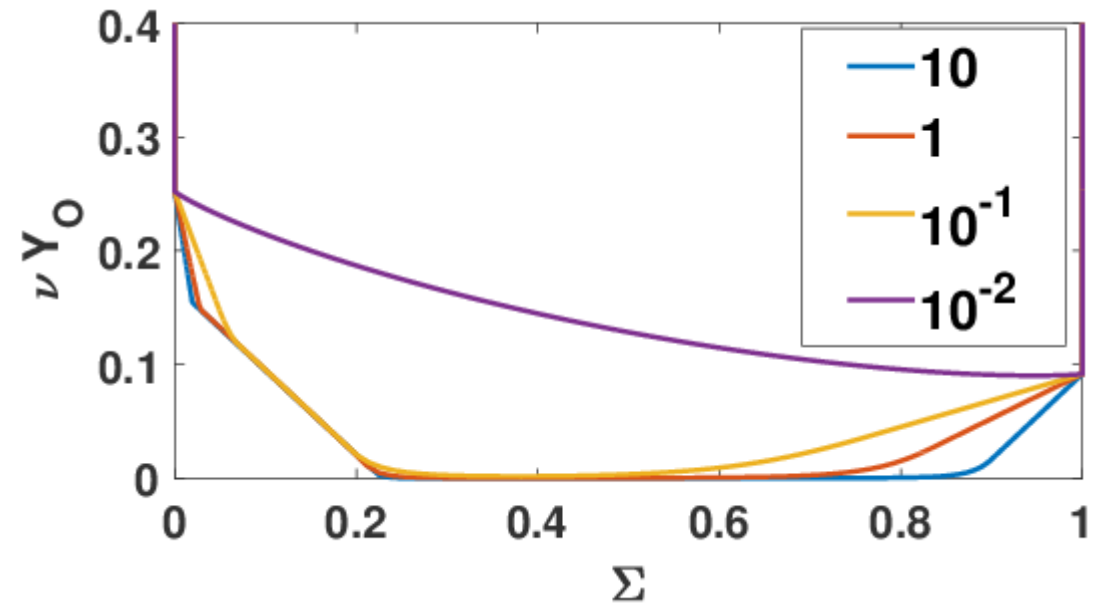
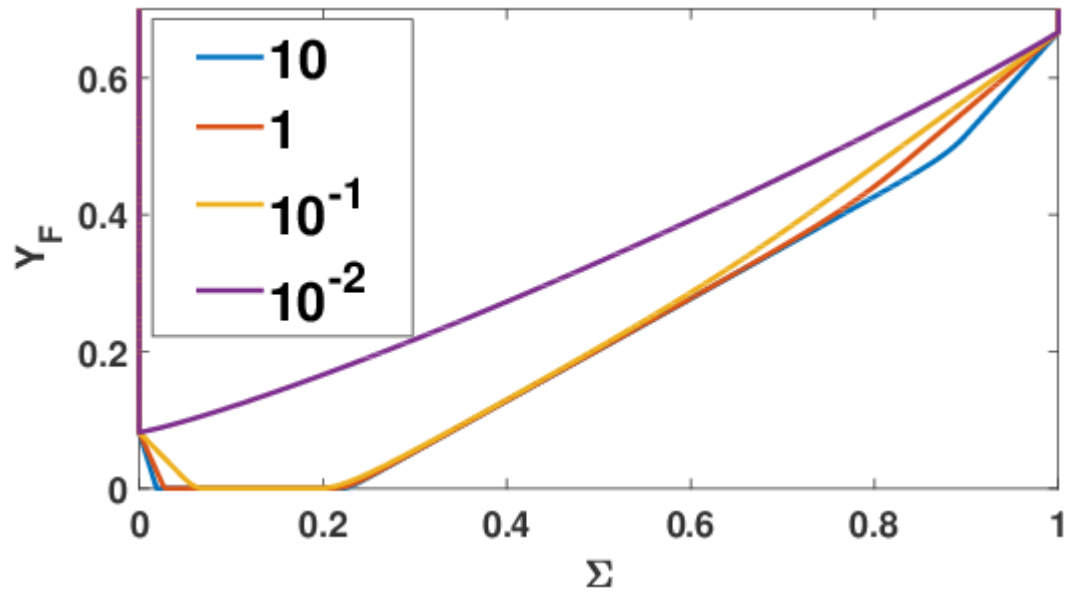
-- For the simple steady-state diffusion-flame-only case, $\Sigma = Z$.

-- At high Da or K values, $n + 1$ linear segments appear for scalar variables versus Σ where n is the number of flames.



-- Consistent results with h behavior are found for fuel and oxygen mass fractions.

-- The scalar dissipation χ is non-zero only in non-linear regions and especially large in the “corner” regions.



Conclusions

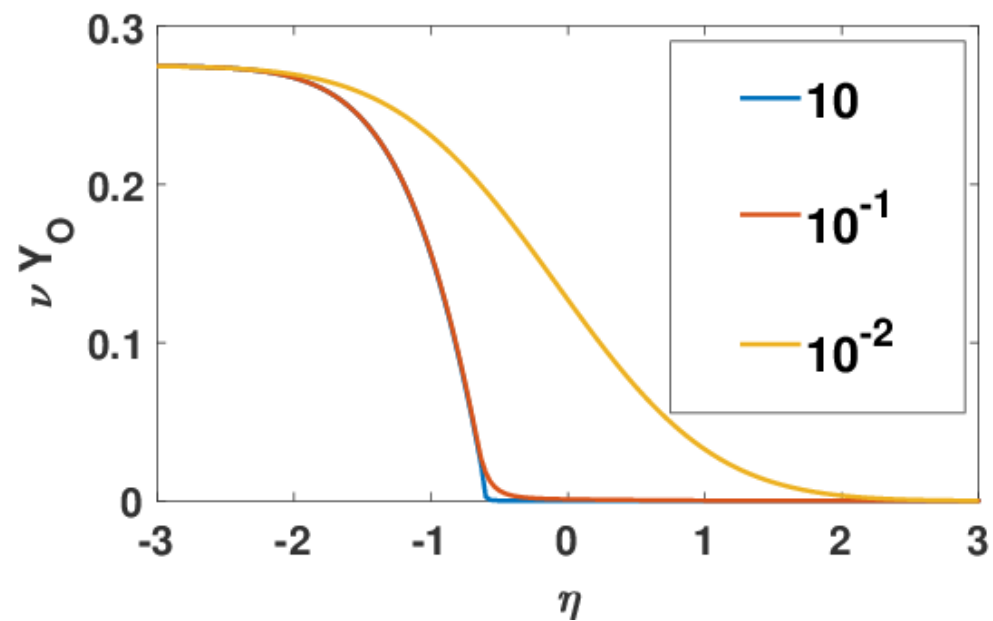
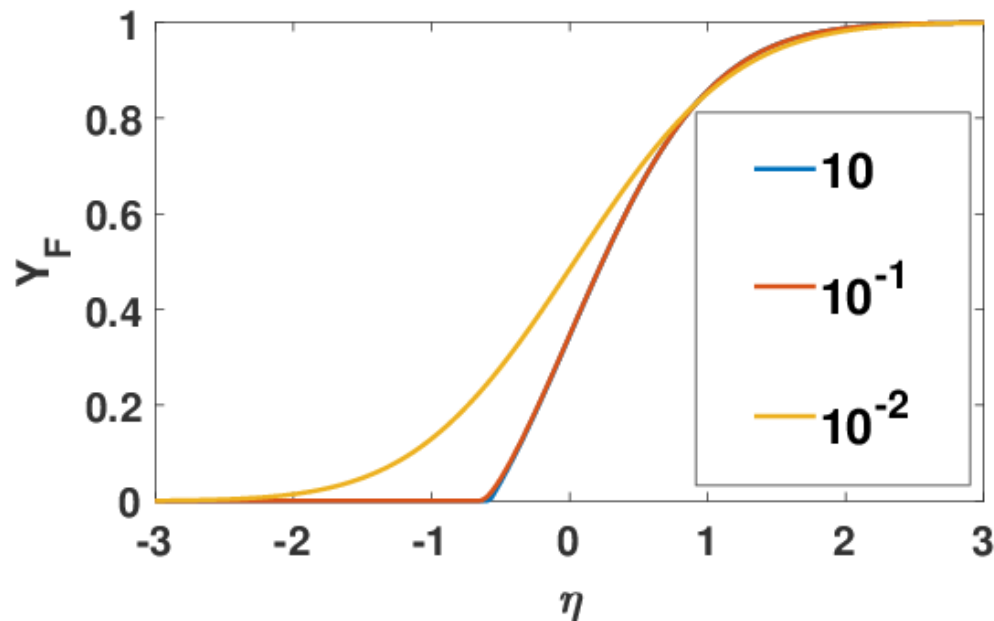
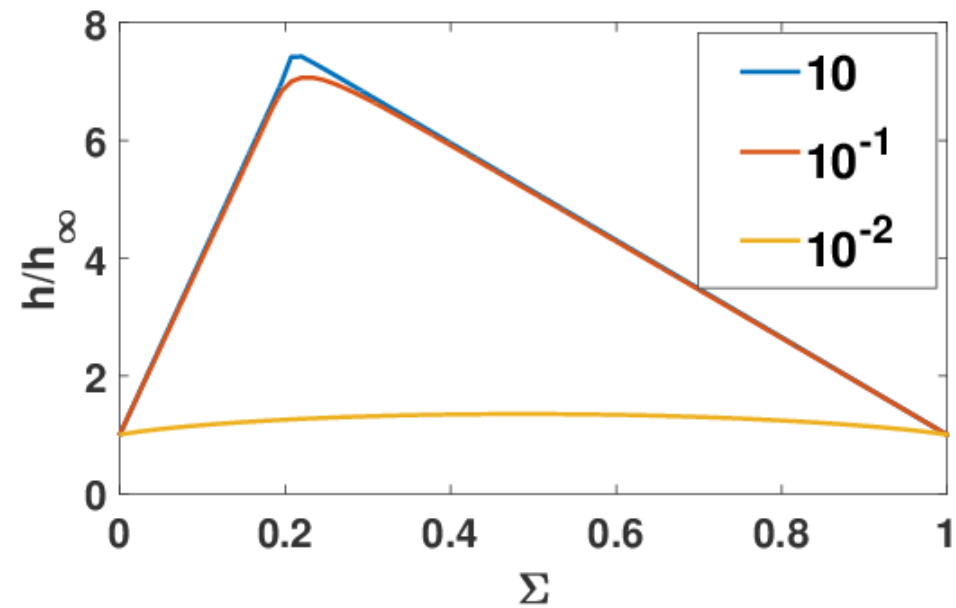
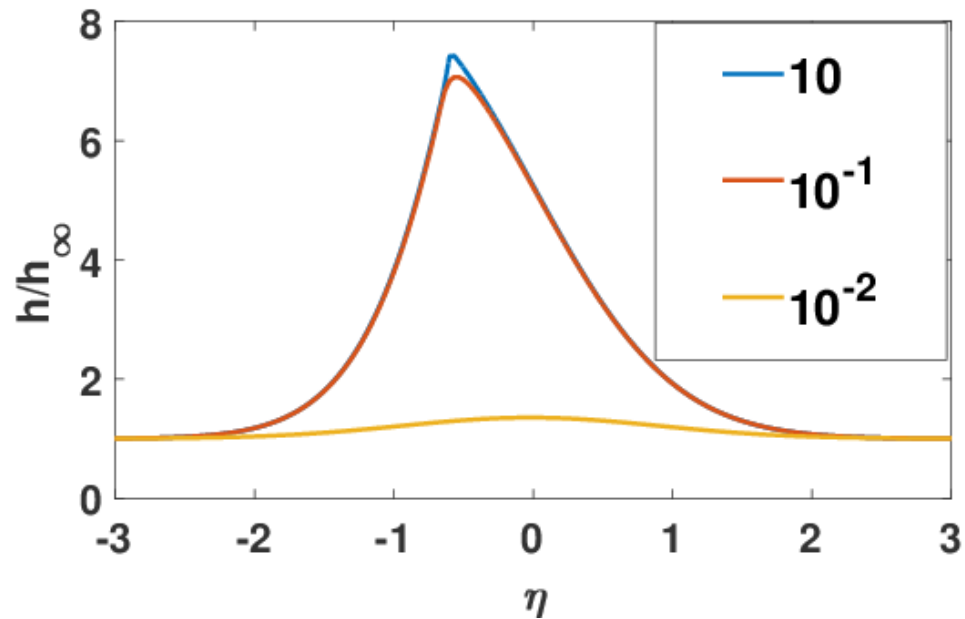
- Flamelet theory has been extended to three dimensions and to multi-flame configurations.
- A formalism has been established unifying various reacting counterflows: a single diffusion flame, a single premixed flame, a two-flame situation with a combustible mixture flowing in one direction, and a three-flame situation with combustible mixtures flowing in both directions.
- Density variations due to combustion result in previously unidentified but substantial velocity overshoots, nonlinear variation in counterflow velocity, and variation in normal strain rate.
- The dependence of multiple-flame existence, flame merging, and extinction on pressure, imposed strain rate, distribution of strain rate, and Prandtl number has been established.

Conclusions continued

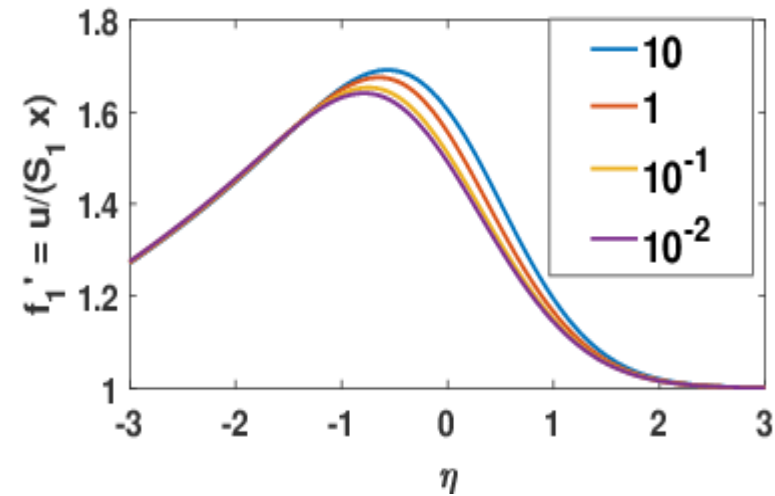
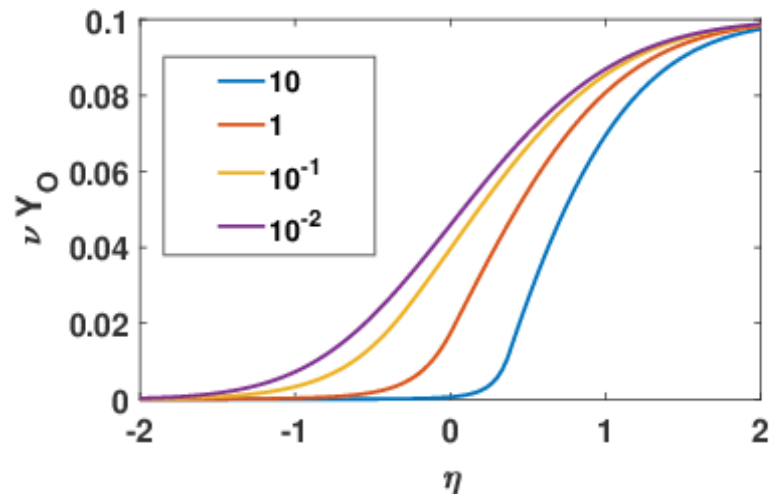
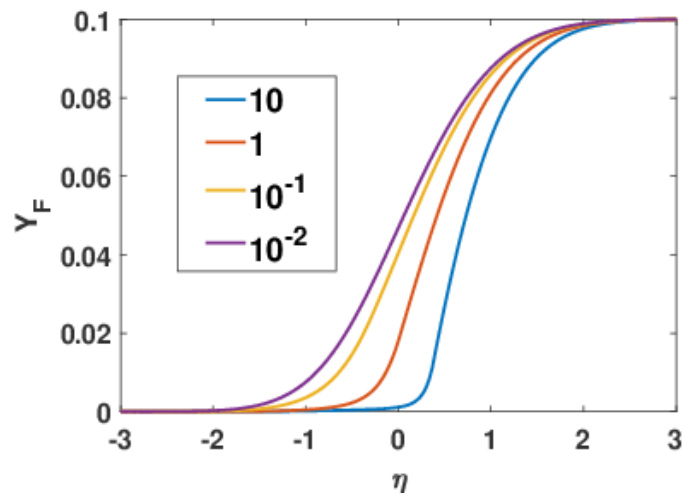
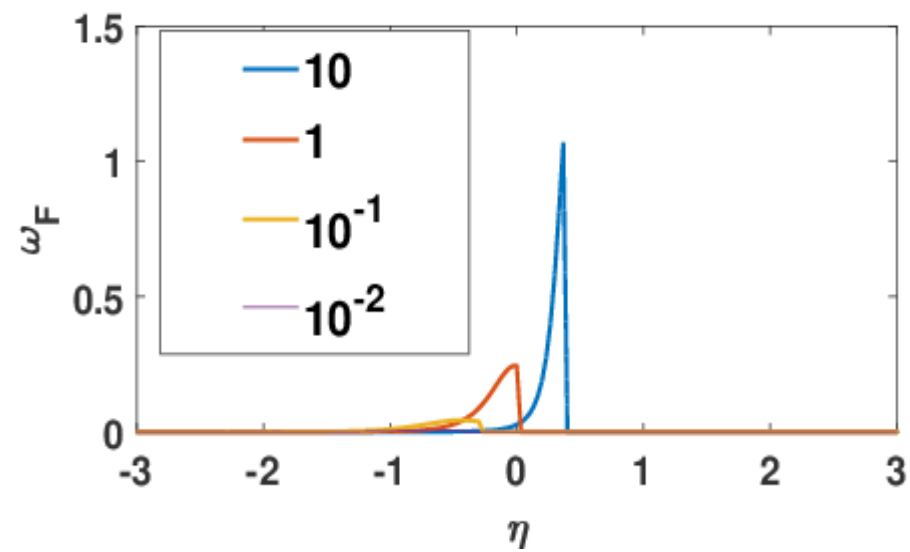
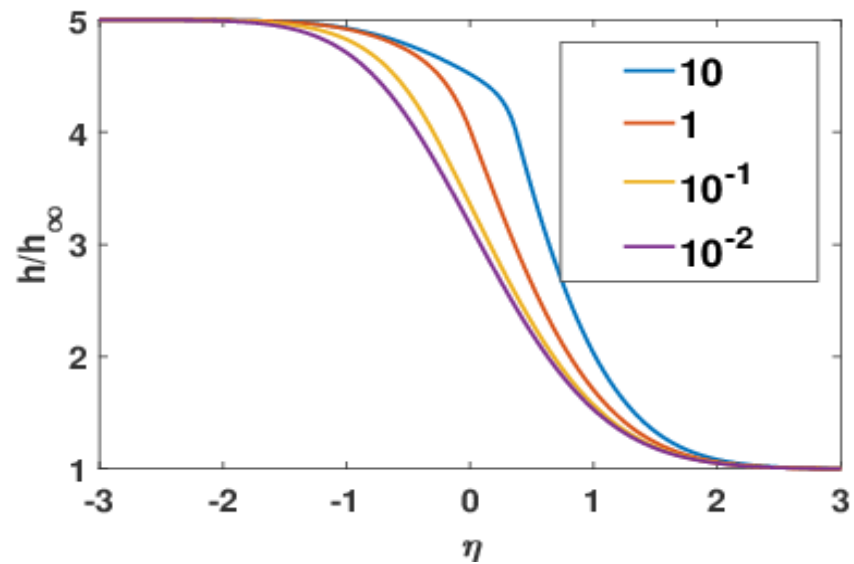
- A generalized variable Σ to replace mixture fraction and a new generalized scalar dissipation rate have been identified.
- Extensions of this flamelet theory for detailed kinetics, detailed transport, and real-fluid equations of state are needed.
- A basis has been provided for development of sub-grid models for LES using the multi-flamelet approach.
- A basis has been provided for further exploration of multi-branched flames in highly strained, three-dimensional flows.

Thank you.

Single Diffusion Flame



Single Premixed Flame



Fuel-rich Mixture flowing against Oxygen

